



National Institute of Information and Communications Technology

*Working Group Meeting on Space-based Lidar Winds*  
(February 7, 2018, NOAA David Skaggs Research Center, Boulder CO, US)

# Recent activities of coherent Doppler Wind Lidar at NICT

**S. Ishii<sup>a</sup>, A. Sato<sup>b, a</sup>, M. Aoki<sup>a</sup>, K. Akahane<sup>a</sup>, S. Nagano<sup>a</sup>, K. Mizutani<sup>a</sup>,  
H. Iwai<sup>a</sup>, K. Okamoto<sup>c, a</sup>, P. Baron<sup>a</sup>, S. Ochiai<sup>a</sup>, and M. Kubota<sup>a</sup>**

a. National Institute of Information and Communications Technology

b. Tohoku Institute of Technology

c. Japan Meteorological Agency / Meteorological Research Institute



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# Outline

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- Background and Objective
- Concept of future space-based Doppler Wind Lidar
- Development of high pulse-energy 2- $\mu$ m laser
- Recent activities at NICT
- Summary

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# Background and Objective

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- Background

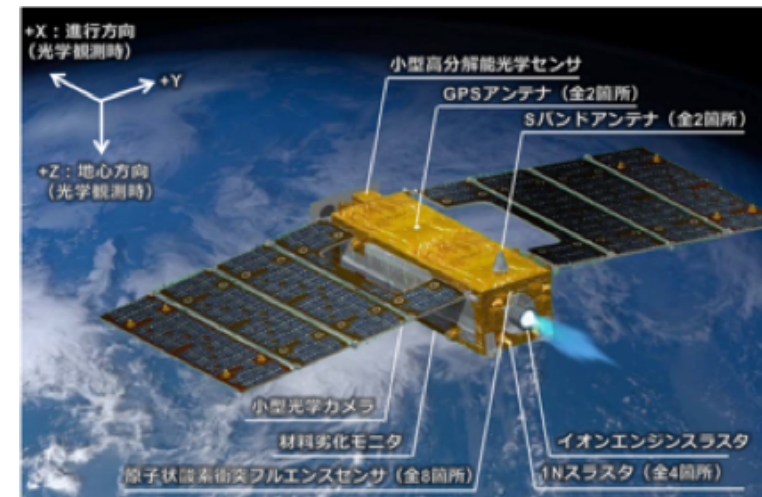
- Global wind profile observation is essential to NWP. Current space-based observing systems are unbalanced toward temperature- and water-vapor-related measurements in comparison with wind measurements.
- Space-based Doppler wind lidar (DWL) is one of promising candidate to fill the gap
- ESA is planning to launch the first space-based DWL Aeolus for global wind profile observations.

- Objectives of this study

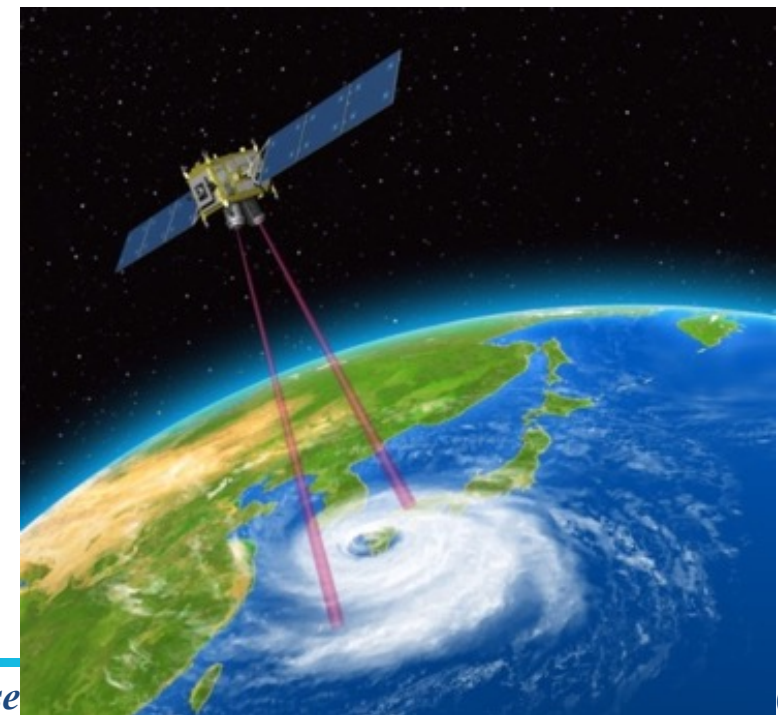
- Development of a reliable stable single-frequency 2- $\mu\text{m}$  pulse laser for DWL
- Demonstration of the 2- $\mu\text{m}$  pulse laser meeting requirements for a future space-based coherent DWL

# Concept of super-low-altitude-satellite-based DWL

	Super Low Altitude space-borne Coherent Doppler Wind Lidar
Orbital altitude Orbit	220 km Polar orbit / Low-inclination orbit
Transmitter	2- $\mu$ m pulse laser (125 mJ, 30 Hz)
Receiver	Heterodyne detection 0.4 m (primary mirror) x 2
Target horizontal resolution	<100 km
Target vertical resolution	Altitude 0-3 km: <0.5 km Altitude 3-8 km: <1 km Altitude 8-20 km: <2 km
Nadir angle	~35 degree
Looking angle	45 and 135 degrees
Target horizontal wind	-100~100 m/sec



[http://www.jaxa.jp/press/2017/10/files/20171027\\_h2af37\\_j.pdf](http://www.jaxa.jp/press/2017/10/files/20171027_h2af37_j.pdf)

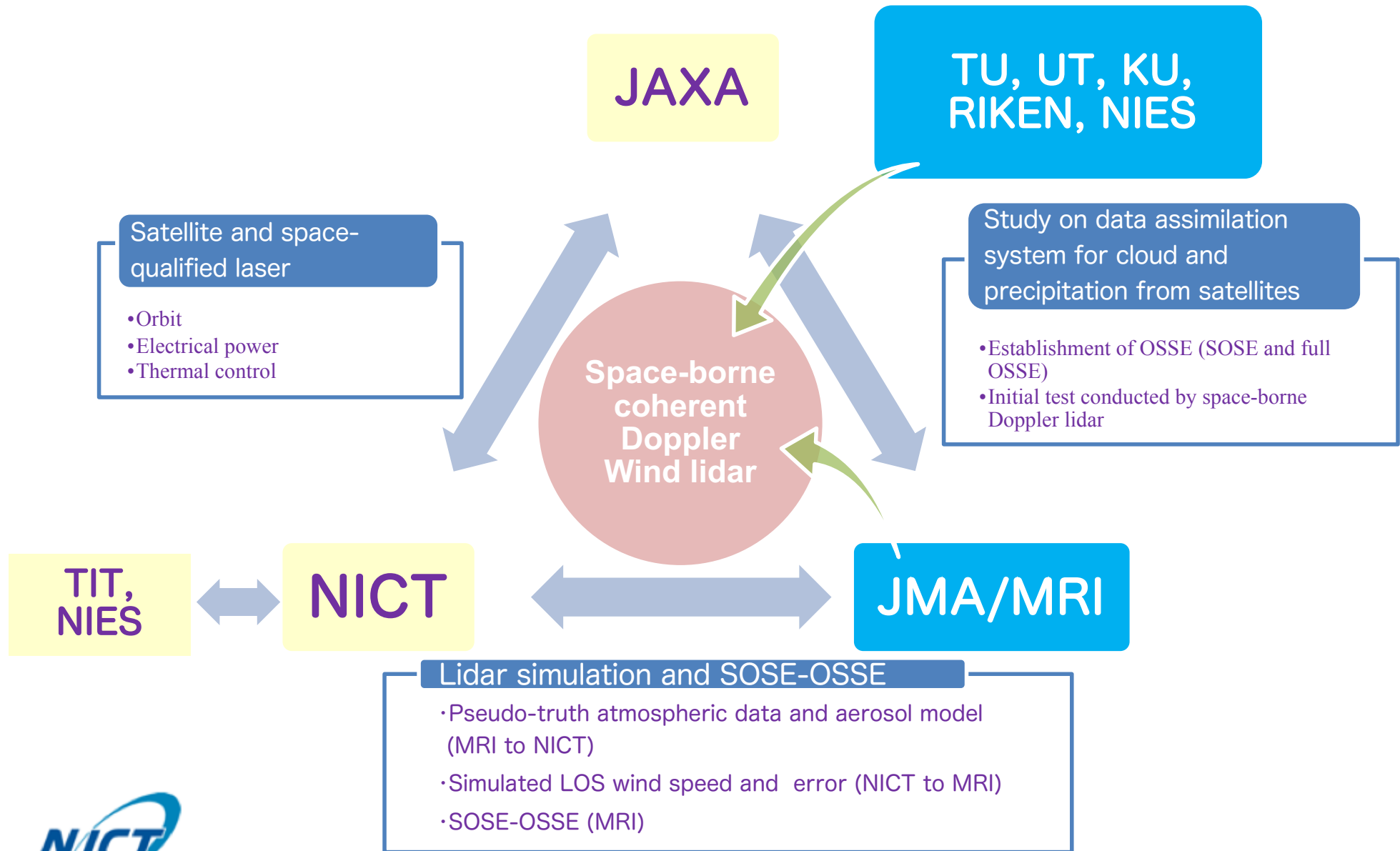


## Super Low Altitude Test Satellite

- December 23, 2017: Launch
- Launch - 3 month: Check
  - Descending: 643 km x 450 km => 392-km circular orbit
- 3 month -
  - Descending: 392 km => 268 km circular orbit
- 15 month - 21 month
  - Keep orbital altitude
  - Ion engine
  - Descending: 392 km => 268 km circular orbit



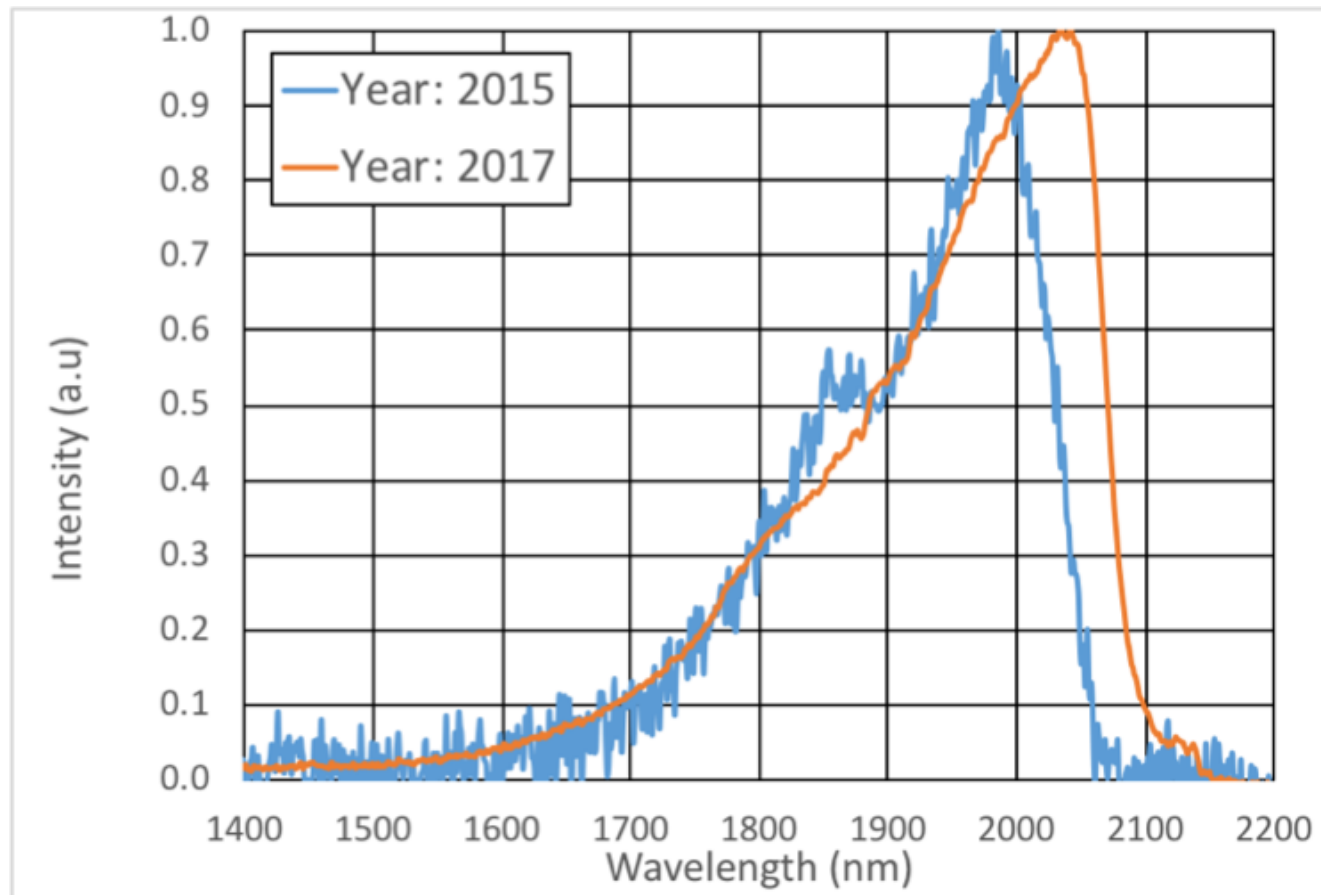
# Collaboration framework for space-borne Doppler lidar



# Development of single-frequency high pulse-energy 2- $\mu$ m laser



# Development of 2- $\mu\text{m}$ semiconductor laser for single-frequency laser



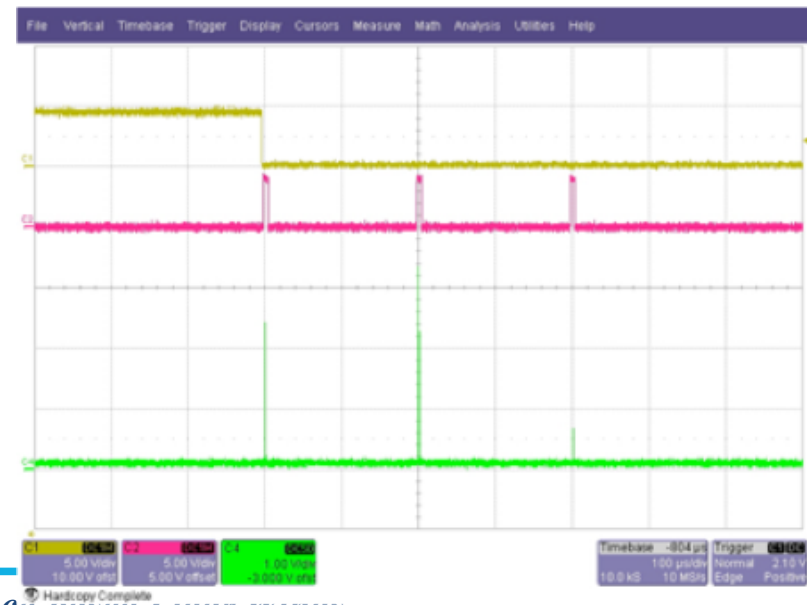
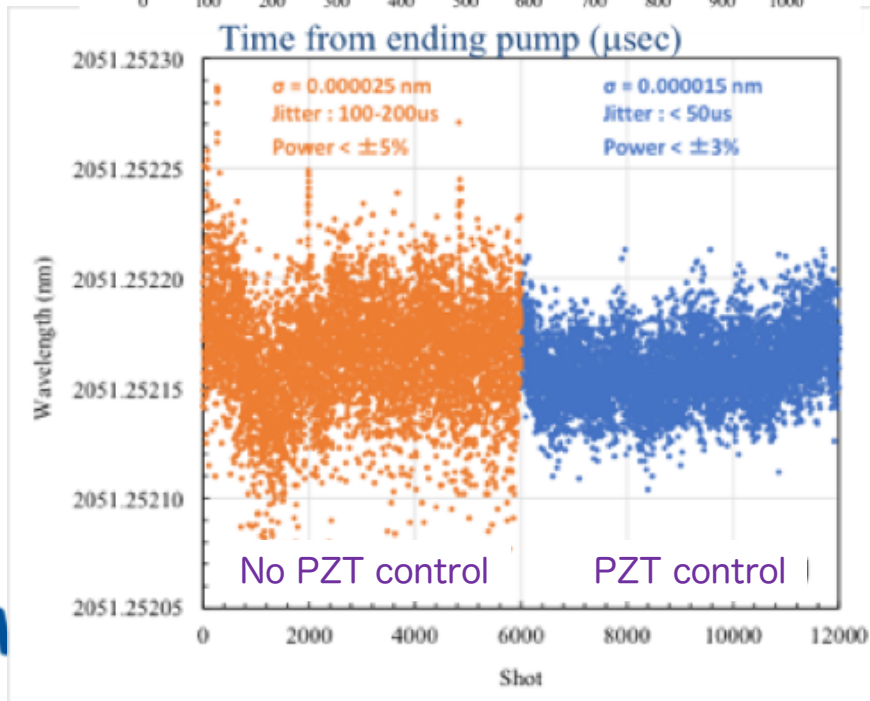
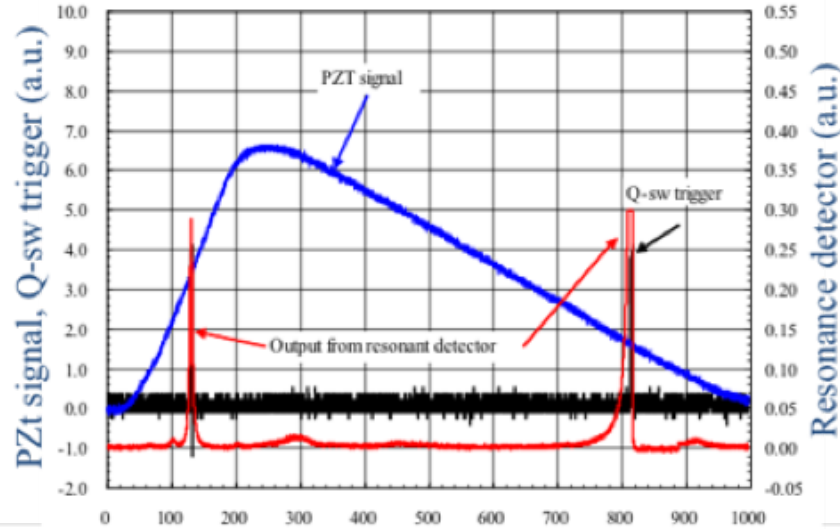
- Highly-stacked InAs quantum-dot layer for 2- $\mu\text{m}$  laser
- Demonstration at 2- $\mu\text{m}$
- Next step
  - High output power
  - Narrow linewidth



# Development of electrical circuit for Laser cavity control

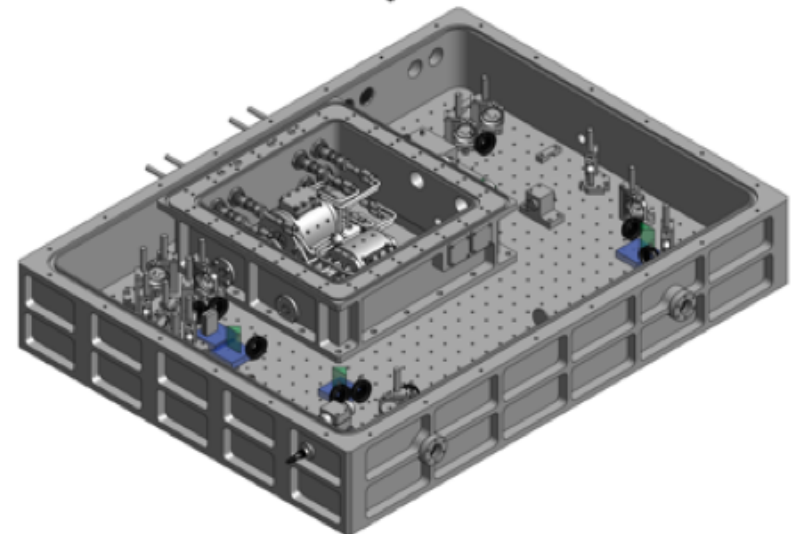
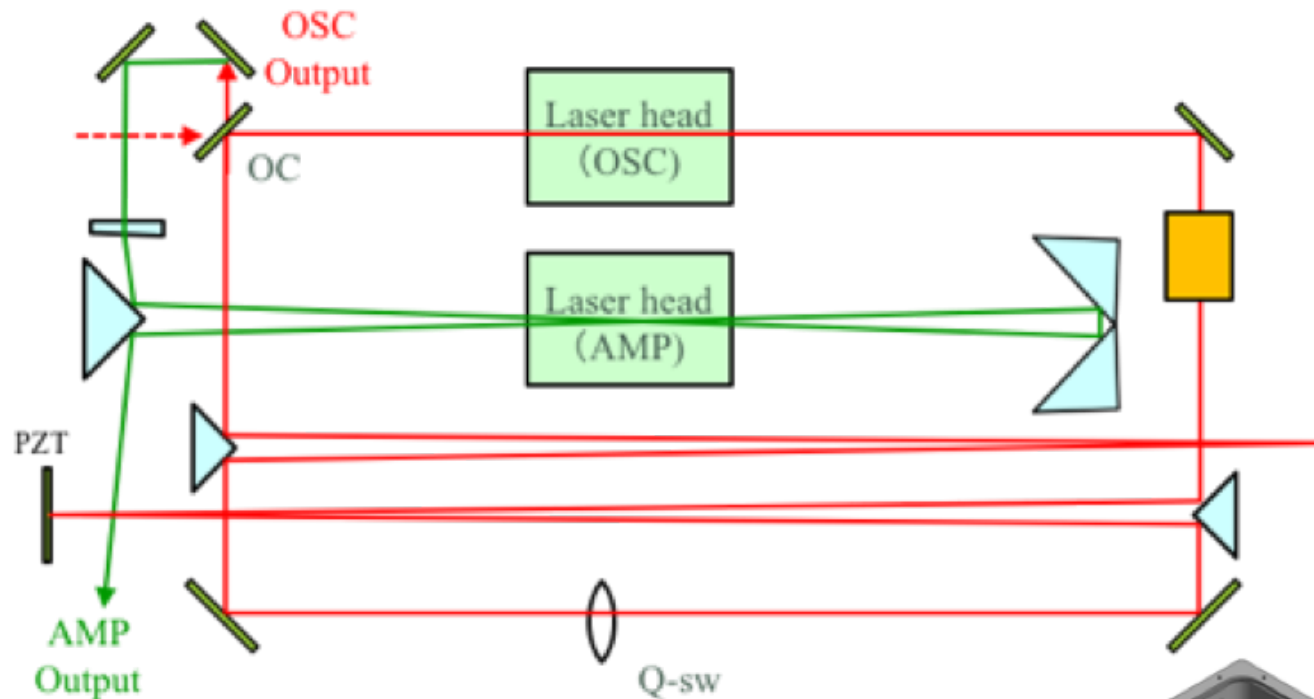
Updating of control circuit for Ramp and fire

Time control for Q-switching for multi pulses





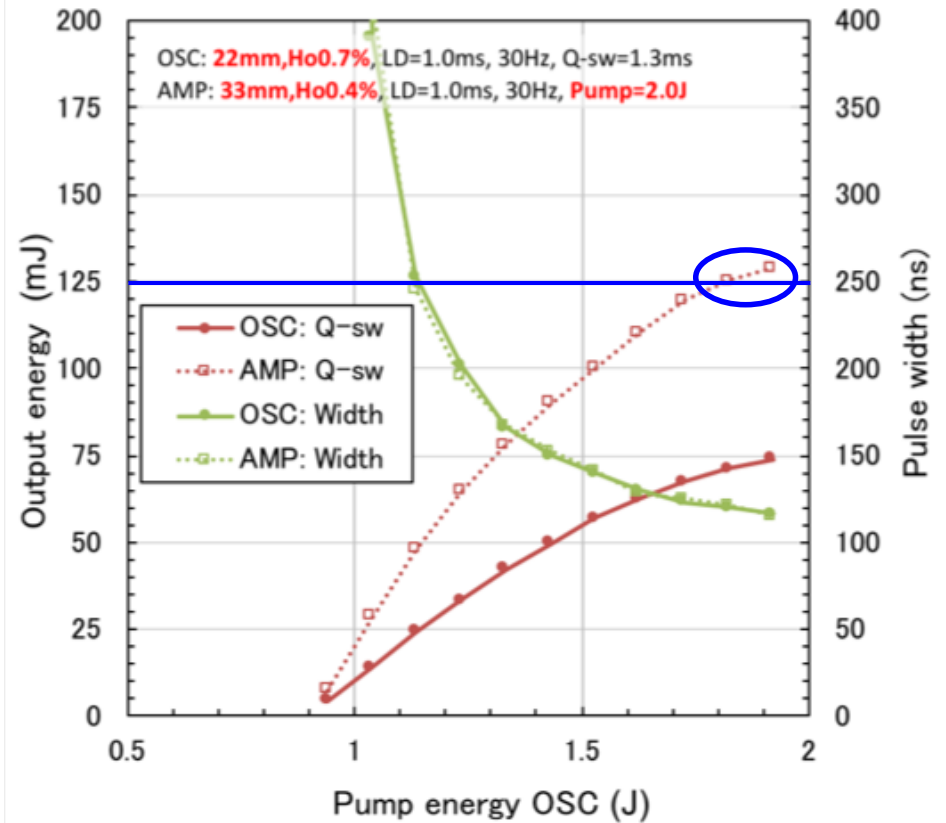
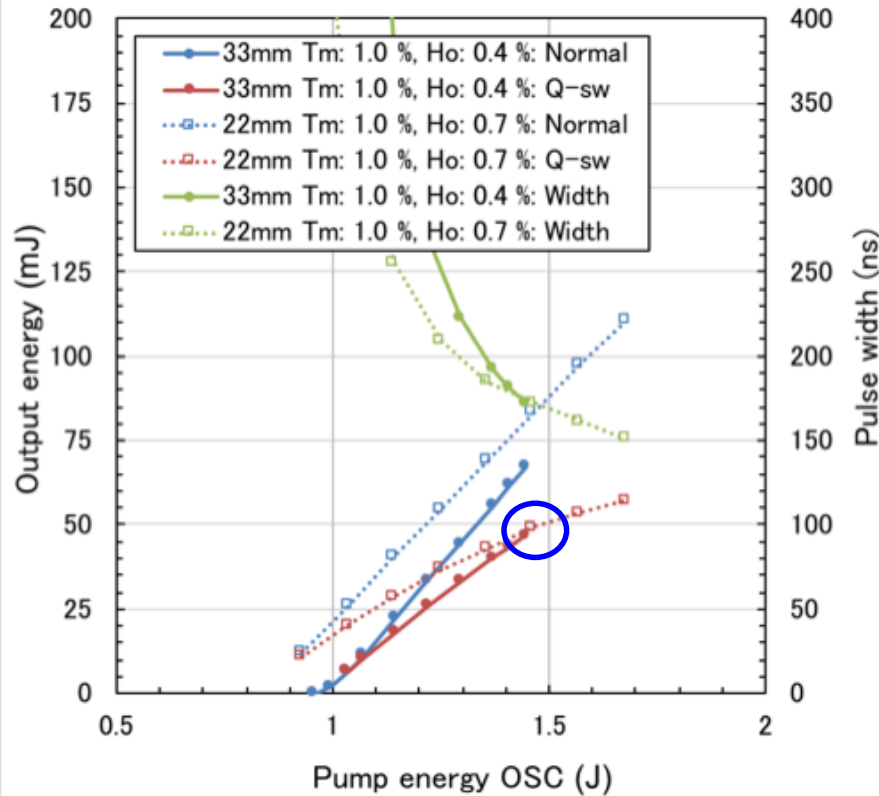
# New 2- $\mu\text{m}$ ring laser



## Design concept

- Simple ring laser design
- Robust and compact ring laser design
- Easy optical alignment

# Lasing characteristics of 2- $\mu\text{m}$ ring laser



OSC (Test I)

Laser rod:  $\phi 4\text{mm} \times 33\text{ mm}$

Tm: 1.0 %, Ho: 0.4 %  $\Rightarrow$  47 mJ at 30 Hz

OSC (Test II)

Laser rod:  $\phi 4\text{mm} \times 22\text{ mm}$

Tm: 1.0 %, Ho: 0.7 %  $\Rightarrow$  50 mJ at 30 Hz

OSC

Laser rod:  $\phi 4\text{mm} \times 22\text{ mm}$

Tm: 1.0 %, Ho: 0.7 %

AMP

Laser rod:  $\phi 4\text{mm} \times 33\text{ mm}$

Tm: 1.0 %, Ho: 0.4 %

## Comparison of high pulse-energy 2-μm laser

FOM	L: 33mm Ho: 0.4% PRF: 30Hz OSC: 1.5J	L: 33mm Ho: 0.4% PRF: 50Hz OSC: 1.5J	L: 22mm Ho: 0.4% PRF: 30Hz OSC: 1.5J	L: 22mm Ho: 0.7% PRF: 30Hz OSC: 1.5J	L: 22mm Ho: 1.0% PRF: 30Hz OSC: 1.5J
Single pulse AMP: x1.5 Single pulse AMP 1.5倍 (1.5J) を仮定	× FOM: 0.087 (64mJ, 50cm) PWR: 240 W WP: 0.8%	○ FOM: 0.113 (64mJ, 50cm) PWR: 400 W WP: 0.8%	○ FOM: 0.110 (80mJ, 50cm) PWR: 240 W WP: 1.0%	◎ FOM: 0.116 (85mJ, 50cm) PWR: 240 W WP: 1.1%	× FOM: 0.085 (62mJ, 50cm) PWR: 240 W WP: 0.8%
Single pulse AMP: x2.0 Single pulse AMP 2.0倍 (2.0J) を仮定	× FOM: 0.104 (76mJ, 50cm) PWR: 280 W WP: 0.8%	○ FOM: 0.135 (76mJ, 50Hz, 50cm) PWR: 467 W WP: 0.8%	○ FOM: 0.129 (94mJ, 50cm) PWR: 280 W WP: 1.0%	○ FOM: 0.110 (100mJ, 50cm) PWR: 280 W WP: 1.1%	× FOM: 0.102 (75mJ, 50cm) PWR: 280 W WP: 0.8%

PWR: Power  
WP: Wall plugin  
Cf.  
Requirements

FOM(Telescope diameter φ40cm, PRF 30Hz):  $0.125\text{J} \times \sqrt{30\text{Hz}} \times (0.4\text{m})^2 = 0.110$

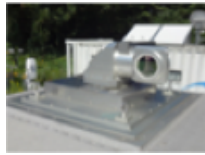
FOM(Telescope diameter φ50cm, PRF 30Hz):  $0.080\text{J} \times \sqrt{30\text{Hz}} \times (0.5\text{m})^2 = 0.110$

FOM(Telescope diameter φ50cm, PRF 50Hz):  $0.062\text{J} \times \sqrt{50\text{Hz}} \times (0.5\text{m})^2 = 0.110$

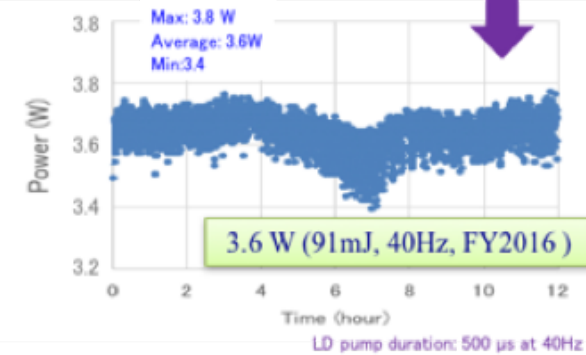
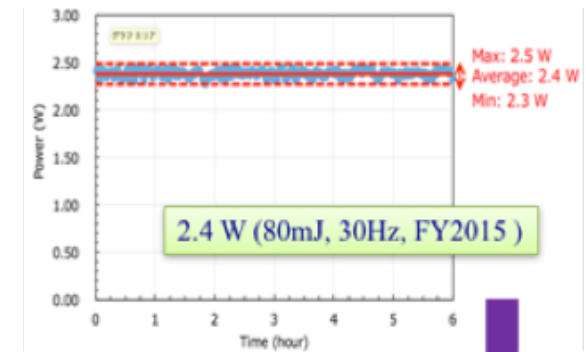
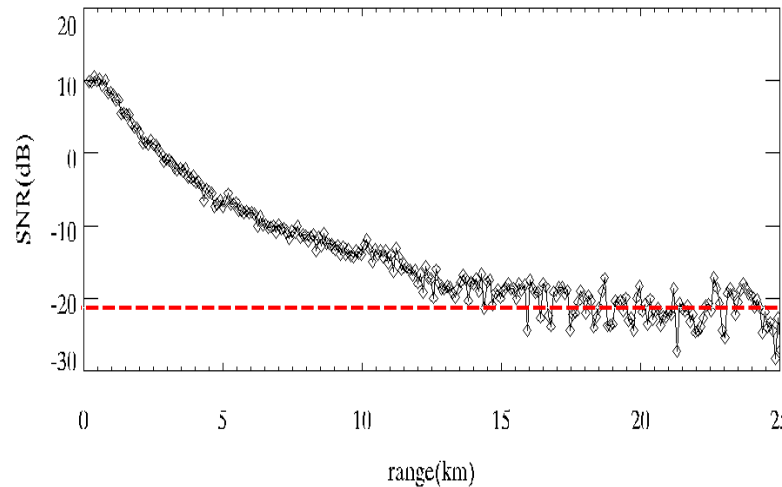
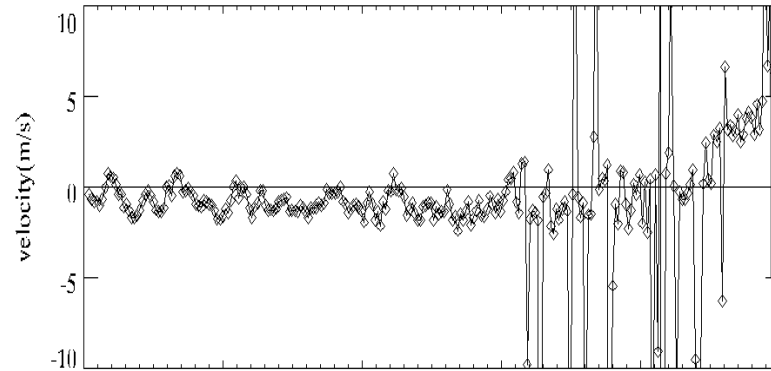
## Recent activities at NICT



# Ground-based 2- $\mu\text{m}$ Doppler Wind lidar



20161014\_151315062\_ 1-sec wind measurement



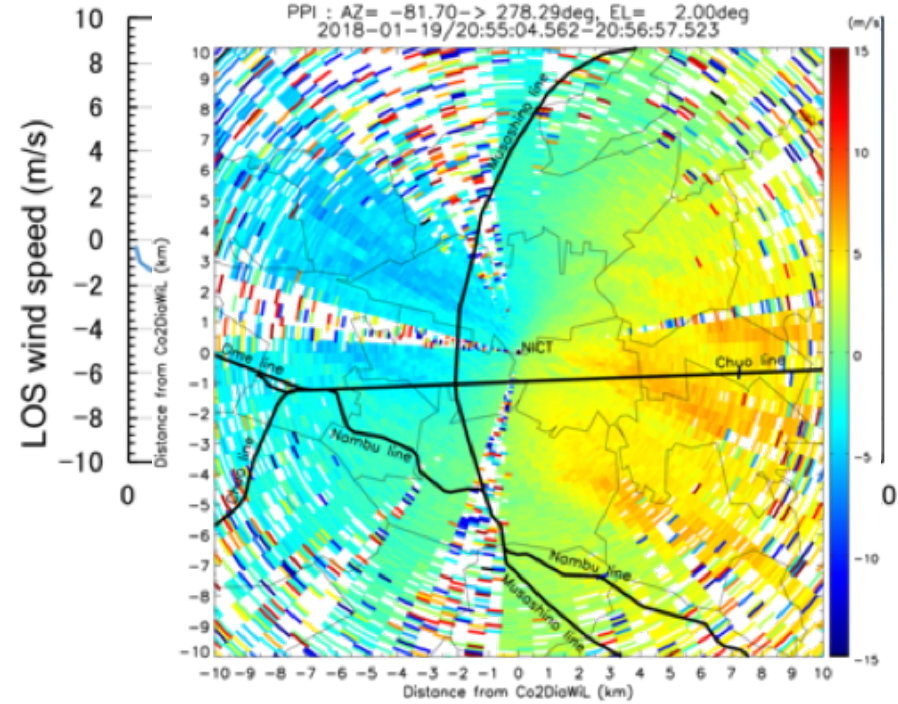
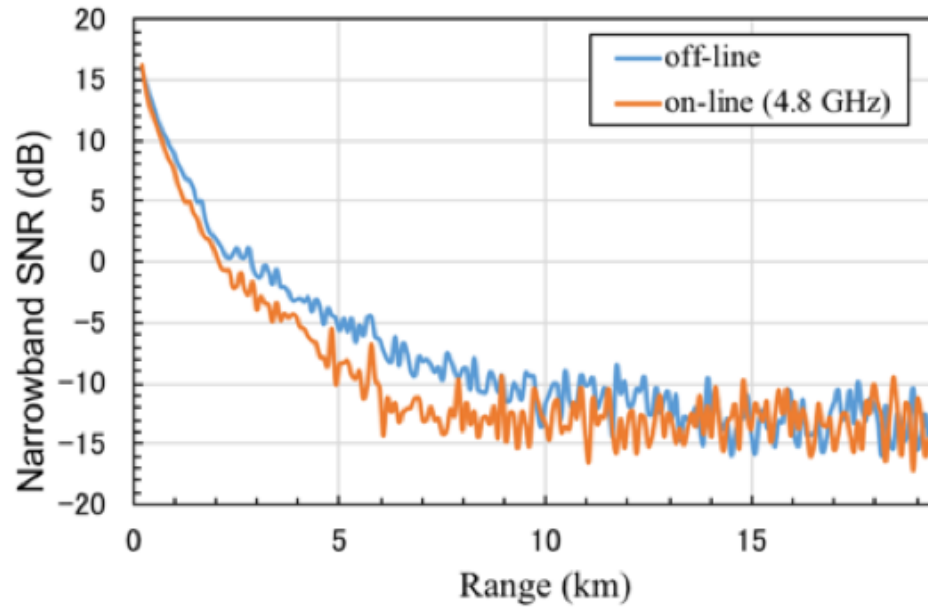
- Optimization
- Pump duration, PRF
  - Focal length of receiver
  - Heterodyne detection



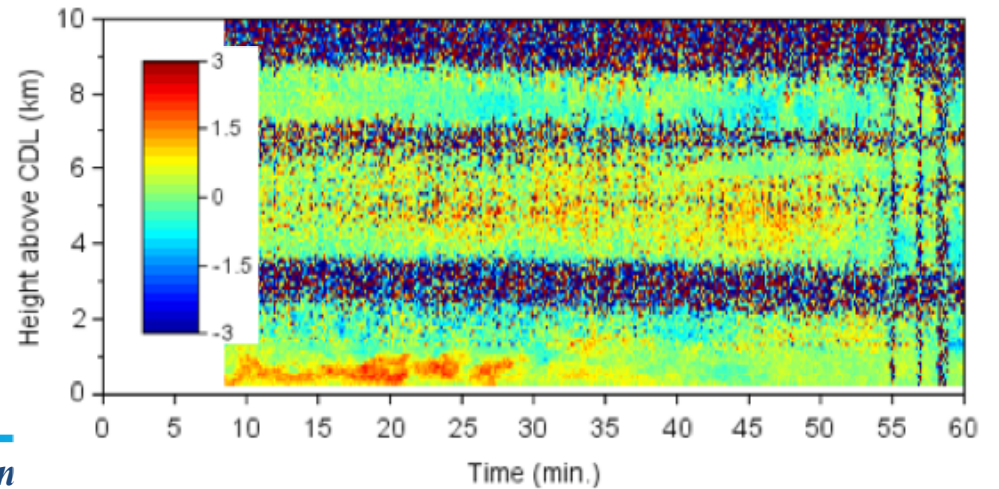


# Airborne 2- $\mu\text{m}$ coherent Lidar for wind and CO<sub>2</sub> measurement

Preliminary result of Airborne 2- $\mu\text{m}$  coherent CO<sub>2</sub>/Doppler Wind Lidar



Radial wind velocity AZ0 EL90 : 2018-01-31 16:08:28 - 2018-01-31 16:59:59





# Validation experiment for the Atmospheric Dynamic Mission-Aeolus

SHOKEN ISHII \*, HIRONORI IWAI \*, MAKOTO AOKI \*, PHILIPPE BARON \*, KOHEI MIZUTANI\*, SEIJI KAWAMURA\*, SATOSHI OCHIAI\*, ANDTOMOAKI NISHIZAWA \*\*

\* NATIONAL INSTITUTE OF INFORMATION AND COMMUNICATIONS TECHNOLOGY

\*\* NATIONAL INSTITUTE FOR ENVIRONMENTAL STUDIES



# Preparation for Aeolus CAL/VAL

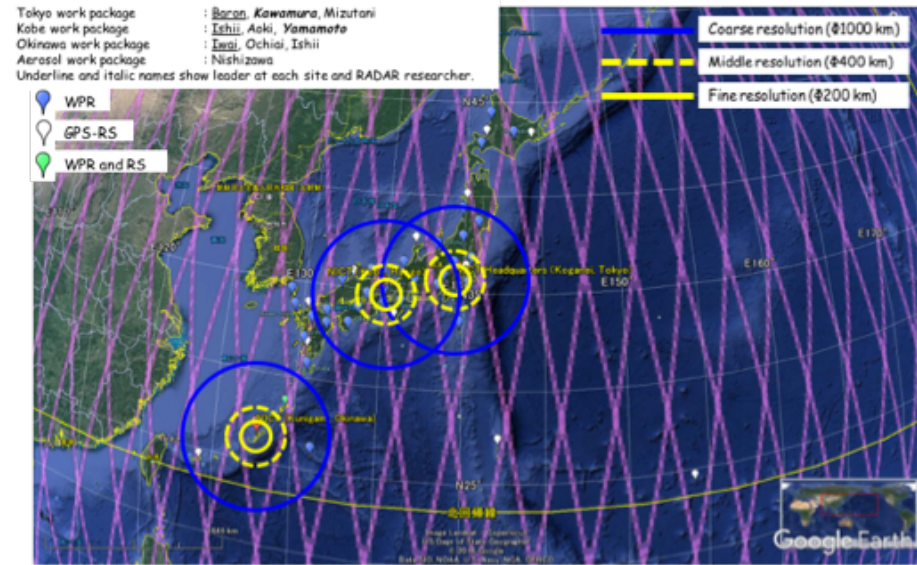
## NICT Tokyo (35.7N, 139.5E)

- Two 2- $\mu\text{m}$  coherent Doppler wind lidars
- 1.3 GHz wind profiler

### VAD



Cols: CO-Is / Baron, Kawamura, Mizutani



## NICT Kobe (34.7N, 135.0E)

- 1.6- $\mu\text{m}$  coherent Doppler wind lidar (WLS400S)



Head and controller for the scanner will be replaced in Kobe.

Cols: CO-Is / Ishii, Aoki, Ochiai

## NICT Okinawa (26.5N, 127.8E)

- 1.6- $\mu\text{m}$  coherent Doppler wind lidar (WLS400S)
- GPS-sonde

Send back to FR in November 2017.



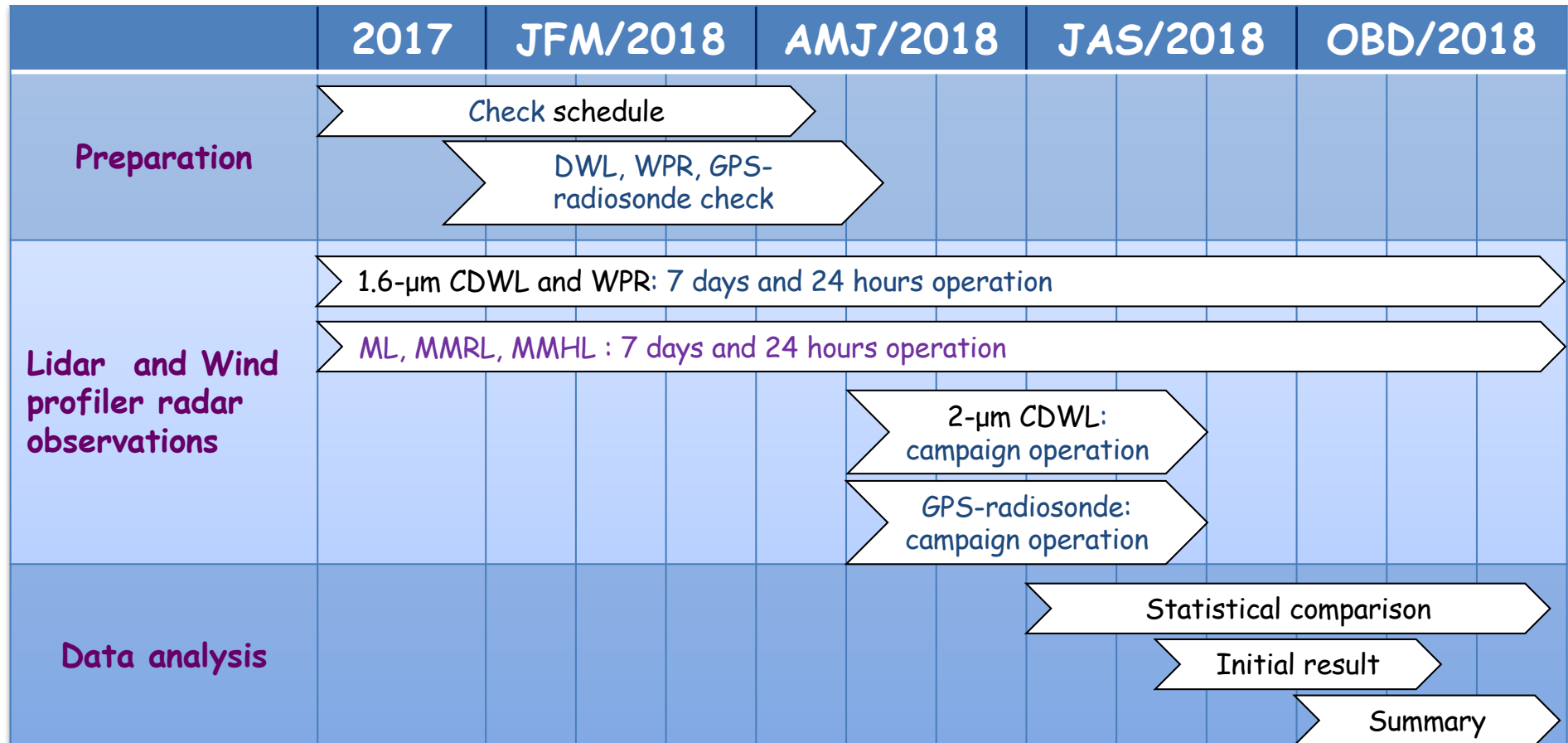
AD, Power supply, EDFA, and circulator were replaced.

Shipped to JA around at the end of this February.

Cols: CO-Is / Iwai, Yamamoto, Ishii



# Schedule



# Validation of the EarthCARE ATLID and MSI products using ground-based lidar and sunphotometry measurements in East Asia.

T. NISHIZAWA\*<sup>1</sup>, A. HIGURASHI\*<sup>1</sup>, R. KUDO\*<sup>2</sup>, H. IRIE\*<sup>3</sup>, K. YASUNAGA\*<sup>4</sup>, M. KATSUMATA\*<sup>5</sup>,  
K. YUMIMOTO\*<sup>6</sup>, S. ISHII\*<sup>7</sup>, H. OKAMOTO\*<sup>6</sup>, K. SATO\*<sup>6</sup>, S. KATAGIRI\*<sup>6</sup>, AND T. Y. NAKAJIMA\*<sup>8</sup>

\*<sup>1</sup>NATIONAL INSTITUTE FOR ENVIRONMENTAL STUDIES, JAPAN, \*<sup>2</sup>METEOROLOGICAL RESEARCH INSTITUTE, JAPAN,  
\*<sup>3</sup>CHIBA UNIVERSITY, JAPAN, \*<sup>4</sup>UNIVERSITY OF TOYAMA, JAPAN, \*<sup>5</sup>JAPAN AGENCY FOR MARINE-EARTH SCIENCE AND  
TECHNOLOGY, JAPAN, \*<sup>6</sup>KYUSHU UNIVERSITY, JAPAN, \*<sup>7</sup>NATIONAL INSTITUTE OF INFORMATION AND  
COMMUNICATIONS TECHNOLOGY, JAPAN, \*<sup>8</sup>TOKAI UNIVERSITY, JAPAN,



# Validation of the EarthCARE ATLID and MSI products using ground-based lidar and sunphotometry measurements in East Asia.

Tomoaki Nishizawa<sup>\*1</sup>, Akiko Higurashi<sup>\*1</sup>, Rei Kudo<sup>\*2</sup>, Hitoshi Irie<sup>\*3</sup>, Kazuaki Yasunaga<sup>\*4</sup>, Masaki Katsumata<sup>\*5</sup>, Keiya Yumimoto<sup>\*6</sup>, Shoken Ishii<sup>\*7</sup>, Hajime Okamoto<sup>\*6</sup>, Kaori Sato<sup>\*6</sup>, Shuichiro Katagiri<sup>\*6</sup>, and Takashi Y. Nakajima<sup>\*8</sup>

<sup>\*1</sup>National Institute for Environmental Studies, Japan, <sup>\*2</sup>Meteorological Research Institute, Japan, <sup>\*3</sup>Chiba University, Japan, <sup>\*4</sup>University of Toyama, Japan, <sup>\*5</sup>Japan Agency for Marine-Earth Science and Technology, Japan, <sup>\*6</sup>Kyushu University, Japan, <sup>\*7</sup>National Institute of Information and Communications Technology, Japan, <sup>\*8</sup>Tokai University, Japan,

## 1. Objectives

The objective of the proposed study is to validate the ATLID L1B, ATLID L2A, MSI L2A, and ATLID-MSI L2B products using ground-based lidar and sunphotometry data, and to contribute to the performance evaluation of EarthCARE observations.

- Main target parameters being Mie co-polar, Rayleigh, and cross-polar attenuated backscatter coefficients at 355 nm (ATLID L1B)
- 10 km-scale aerosol-oriented 355 nm extinction, backscatter, and depolarization profiles (A-AER/ATLID L2A)
- 355 nm cloud and aerosol extinction, backscatter, and depolarization profiles (A-EBD/ATLID L2A)
- aerosol layer products (A-ALD/ATLID L2A)
- aerosol optical thicknesses (AOTs) at 670 and 865 nm (M-AOT/MSI-L2A)
- columnar aerosol optical properties (AM-ACD/ATLID-MSI L2B).

# Summary

Recent research and activities of 2- $\mu\text{m}$  coherent Lidar at NICT are reported in the presentation:

- Development of a single-frequency 2- $\mu\text{m}$  high pulse-energy laser meeting the requirements for the space-based CDWL
- Development of single-frequency semiconductor laser (on going)
- Q-switched output pulse energy of 125 mJ operating at 30 Hz at a laser rod temperature of -40C
- Preliminary results of airborne 2- $\mu\text{m}$  coherent lidar
- Preparation for CAL/VAL activities for ESA missions: Aeolus and EarhCARE (on going)

Future works:

- Development of single-frequency semiconductor laser
- Demonstration of a single-frequency semiconductor laser with optical fiber amplifier
- Optimization of performance of a new 2- $\mu\text{m}$  ring laser
- Environment test

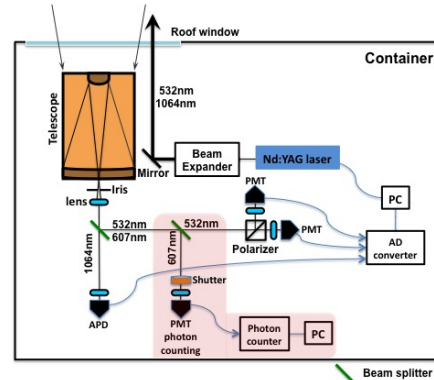


**Back up**

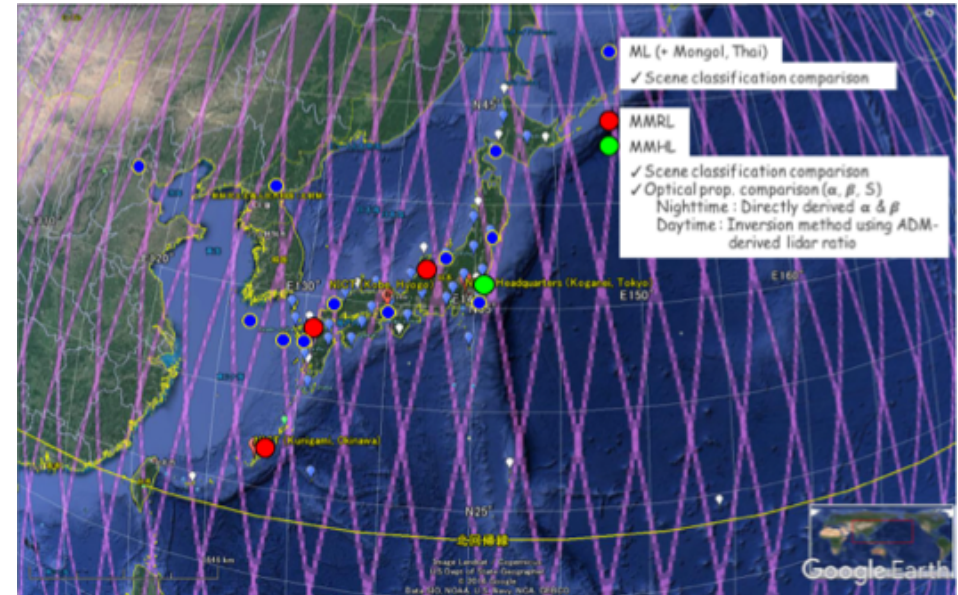


# NIES aerosol profiling

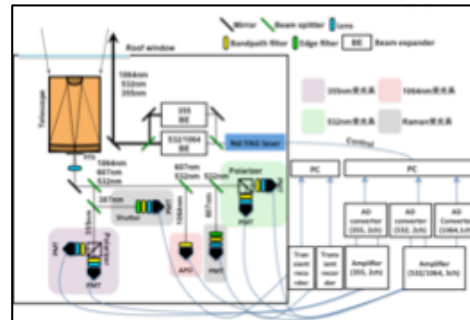
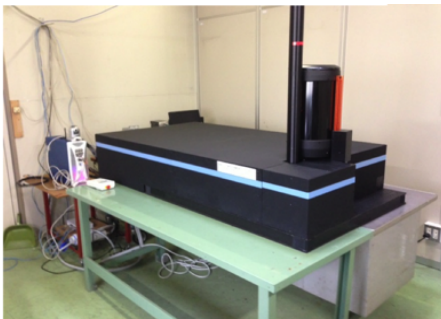
## ● Dual-wavelength polarization Mie lidar (ML) ( $2\beta(532,1064)+1\delta(532)$ )



\*24-hour continuous operation

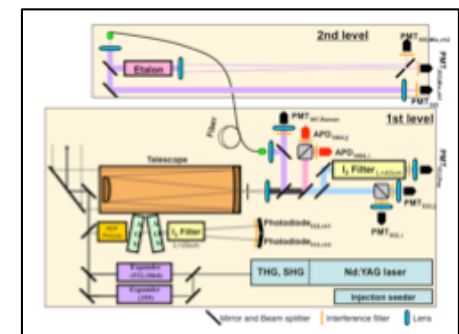


## ● Multi-wavelength Mie-Raman lidar (MMRL) ( $2\alpha(355,532) + 3\beta(355,532,1064)+2\delta(355,532)$ )



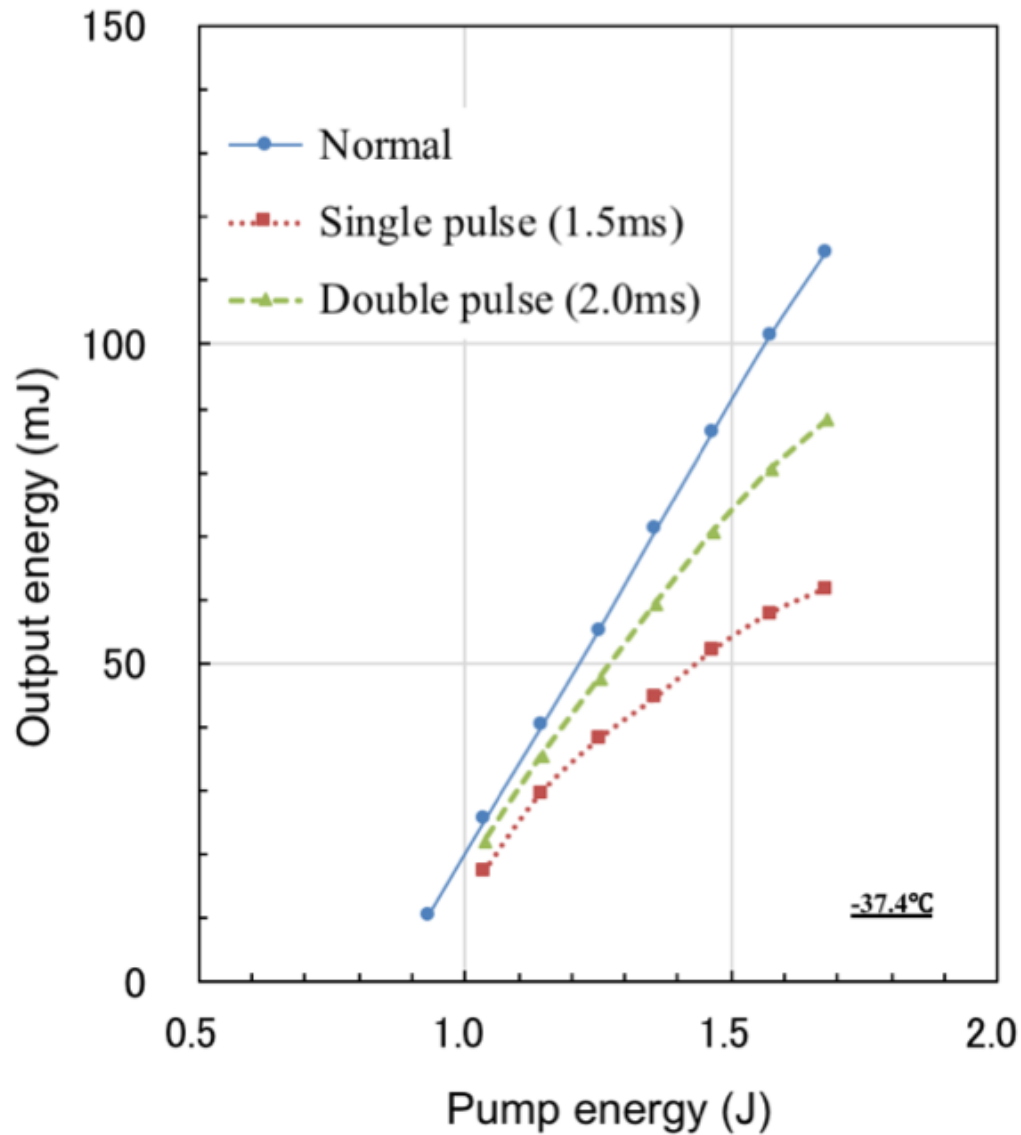
\*24-hour continuous operation

## ● Multi-wavelength Mie-HSRL ( $2\alpha(355,532) + 3\beta(355,532,1064)+2\delta(355,532)$ )



\*24-hour continuous operation except for 355 HSRL

# Lasing characteristics of 2- $\mu\text{m}$ ring laser: double pulse



OSC

Laser rod:  $\phi 4\text{mm} \times 22\text{ mm}$

$T_m$ : 1.0 %,  $H_o$ : 0.7 %

AMP

Laser rod:  $\phi 4\text{mm} \times 33\text{ mm}$

$T_m$ : 1.0 %,  $H_o$ : 0.4 %

# Tm-doped fiber amplifier for single-frequency 2- $\mu\text{m}$ CW laser

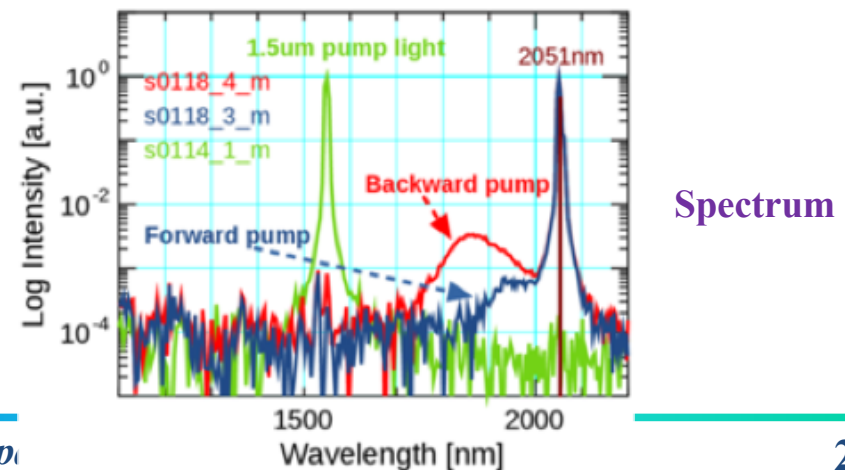
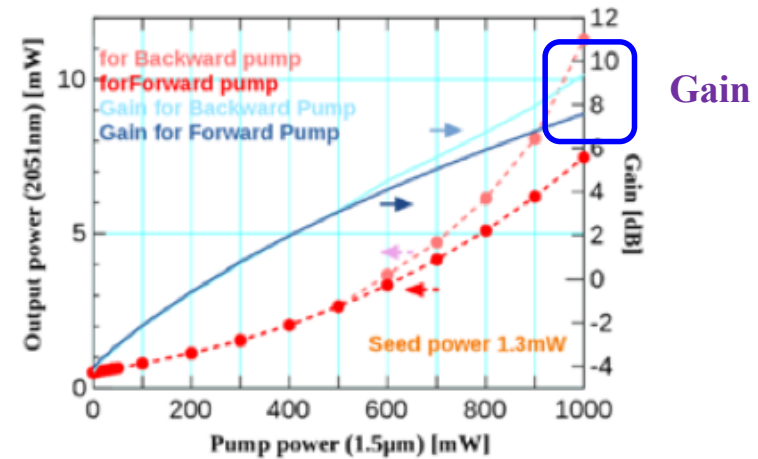
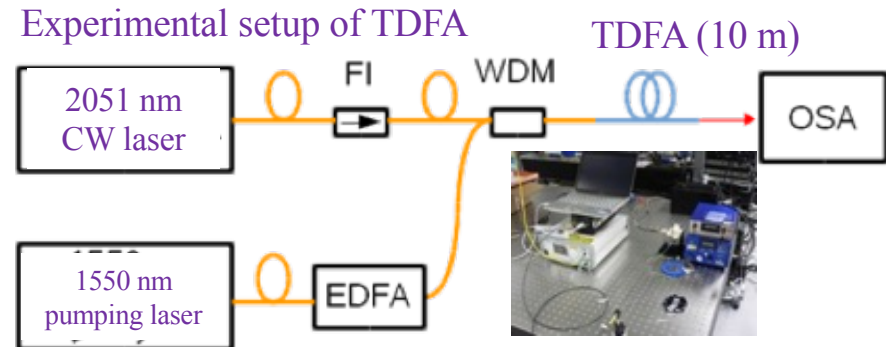
- Requirement for optical output power
  - Wavelength: 2051 nm
  - Gaining: 13 dB (input optical power: ~1mW)
  - Output: 20 mW
- Other requirements
  - Long-term mechanical stability (vibration and shocks)
  - Maintenance-free
  - Compact and high efficiency (E-O conversion, removal heat)
  - Space-qualification

Erbium-Doped Fiber AMP: EDFA



Thulium-Doped Fiber for space application

- Double-clad Tm optical fiber
- 793nm multi-mode LD pumping
- All optical fiber



# 2- $\mu\text{m}$ laser technology required for space-based CDWL

Technology required for space-borne CDWL		Task
2- $\mu\text{m}$ single-frequency Q-switched laser	Single-frequency 2- $\mu\text{m}$ CW laser for seeding technique	<ul style="list-style-type: none"> <li>● High power</li> <li>● High stability</li> <li>● Narrow line width</li> <li>● Availability</li> </ul>
	High pulse-energy laser	<ul style="list-style-type: none"> <li>● High pulse energy</li> <li>● High stability</li> <li>● High efficiency</li> </ul>
Detector		<ul style="list-style-type: none"> <li>● High-speed response</li> <li>● Low NEP</li> </ul>

- Current Tm,Ho:YLF laser system: 80mJ  $\times$  30Hz.
- Laser rod temperature: -80  $^{\circ}\text{C}$ 
  - Power consumption of the cooling system

Target pulse energy: **125 mJ x 30 Hz**

